

CRITICAL ITEMS LIST (CIL)

No. 10-01-01-10R/01

SYSTEM: Space Shuttle RSRM 10 CRITICALITY CATEGORY: 1 SUBSYSTEM: Case Subsystem 10-01 PART NAME: Factory Joint, Moisture Seal and ASSEMBLY: Case 10-01-01 Thermal Protection System (TPS) (1) 10-01-01-10R Rev M FMEA ITEM NO.: PART NO.: (See Section 6.0) Boost (BT) CIL REV NO.: PHASE(S): М DATE: 31 Jul 2000 QUANTITY: (See Section 6.0) SUPERSEDES PAGE: EFFECTIVITY: (See Table 101-6) 209-1ff. 30 Jul 1999 HAZARD REF.: BC-11 DATED: CIL ANALYST: D. J. McGough APPROVED BY: DATE: RELIABILITY ENGINEERING: K. G. Sanofsky 31 Jul 2000 S. R. Graves ENGINEERING: 31 Jul 2000 1.0 FAILURE CONDITION: Failure during operation (D) 1.0 Moisture seal/TPS fails to protect factory joint from aerodynamic heating 2.0 FAILURE MODE: 3.0 FAILURE EFFECTS: Debris would damage the ET and orbiter. Loss of tension in pin retention strap due to overheating may let the pins work loose. This could cause a joint failure causing loss of the RSRM, SRB, crew, and vehicle 4.0 FAILURE CAUSES (FC): FC NO. DESCRIPTION FAILURE CAUSE KEY 1.1 Nonconforming raw material properties Α 1.2 Voids or inclusions В 1.3 Bondline failure 1.3.1 Contamination of bonding enhancement material С Bonding surface not adequately cleaned 1.3.2 D 1.3.3 Bonding enhancement material not properly applied Ε 1.3.4 F Vacuum bag leaks 1.4 Nonconforming cure G 1.5 Nonconforming physical and mechanical properties Η Transportation and handling damage 1.6 1.7 Age degradation 1.8 Moisture or fungus degradation Κ 1.9 Ablative failure under aerodynamic loads

REVISION M

DOC NO.	TWR-157	VOL	II	
SEC	209	PAGE	1	



DATE: SUPERSEDES PAGE: DATED:

31 Jul 2000 209-1ff. 30 Jul 1999

1.10 Acoustic vibration and aeroshear

M

5.0 REDUNDANCY SCREENS:

SCREEN A: N/A SCREEN B: N/A SCREEN C: N/A

6.0 ITEM DESCRIPTION:

 This CIL addresses aerodynamic heat failure of the external factory joint moisture seal during boost phase of flight. The factory joint seal is applied in casting segment assemblies:

1U76666	Case Assembly, Forward Segment Insulated
1U76667	Case Assembly, Center Segment Insulated
1U77503	Case Assembly, Aft Segment Insulated

- 2. External factory-joint weather seals consist of a specially formulated, silica-filled, ethylene propylene hexadiene monomer (EPDM-neoprene) rubber mixture, extruded or calendared to specific dimensional requirements per engineering.
- Extruded stocks used on the external factory joint seal are produced in three specific sizes or "types" as follows:

EPDM, Extrusion or EPDM, Silica-Filled, Uncured Type I Figure 1 EPDM, Extrusion or EPDM, Silica-Filled, Uncured Type II Figure 2 EPDM, Extrusion or EPDM, Silica-Filled, Uncured Type III Figure 3

Type I is used everywhere circumferentially around the factory joint except under the systems tunnel. The Type I seal is approximately 6-inches wide, roughly one-third of an inch thick, and 6 3/4-ft long.

- 4. Calendared stock is produced in four nominal thickness ranges with laminated thickness and/or number limits per engineering. Only the 0.060-inch thick size is used on the external factory joint seals to overlap Type I and II butt joints.
- 5. Figure 4 provides radial placement information at each joint to show where each type of seal is used. Figure 5 provides cross-sectional view information on how each type of seal overlaps and fits the segment clevis and tang joint. Figure 6 depicts calendared stock use at extruded butt joints.
- 6. Primer and an adhesive are necessary to apply an external factory joint seal. Materials are listed in Table 1.

DOC NO. TWR-15712 | VOL | II | SEC | 209 | PAGE | 2



DATE: SUPERSEDES PAGE: DATED: 31 Jul 2000 209-1ff. 30 Jul 1999

TABLE 1. MATERIALS

Drawing No. Name	Material	Specification	Quantity
Hatband Joint Extrusion, EPDM Rubber	Silica-Filled EPDM	STW4-3775 Type I STW4-3775 Type VIII STW4-3775 Type IX STW4-3775 Type X	46 lb 13 ea 2 ea 6 ea
Thermal Insulation, EPDM Rubber	Silica-Filled EPDM	STW4-2536 or STW4-3775	14 A/R
Adhesive Primer, Rubber-To-Metal Adhesive, Rubber-to-Metal, Elevated Temp	Chlorinated Rubber Chlorinated Rubber	STW5-2664 STW5-2798	A/R A/R

6.1 CHARACTERISTICS:

- EPDM terpolymers are known for their extreme weather-resistant properties and ability to be extended (capacity for absorbing filling agents). The EPDM capacity for absorbing silicone additives gives the seal greater strength at high temperatures needed for external seal application to combat effects of aerodynamic heating. It also exhibits excellent low temperature flexibility.
- Approximately 20 percent of the seal formula contains neoprene rubber (chlorobutadiene) that exhibits excellent moisture rejection, shape-holding ability, flexibility at high temperatures, and ozone resistance.
- 3. External factory joint seals are not cork covered as are the field joints. Therefore, the EPDM-neoprene rubber seal must also act as the thermal protection system (TPS) for the factory joint. The external EPDM-neoprene rubber factory joint seal must act as a total environmental protection system of the factory clevis and tang mechanical joint.
- 4. Extruded EPDM seals are shaped to fit across the external factory joint lip. Enough overlap is designed into the seal to assure the bondline is maintained through all phases of launch and recovery, and that during vulcanization, enough surface area exists on each side of the joint to achieve good rubber-to-metal bonding.

Seals should:

- a. Provide moisture protection of joint components
- b. Prevent entry of any liquids such as rain
- c. Provide thermal protection of joint components during the boost phase
- d. Maintain sealing integrity through mechanical shock, vibration, and expansion loads occurring during the boost phase
- e. Survive aerodynamic heating effects and ablative wear associated with the boost phase

7.0 FAILURE HISTORY/RELATED EXPERIENCE:

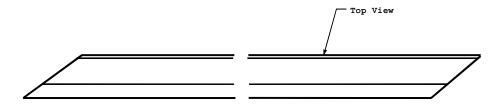
 Current data on test failures, flight failures, unexplained failures, and other failures during RSRM ground processing activity can be found in the PRACA database.

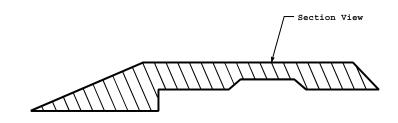
8.0 OPERATIONAL USE: N/A

DOC NO.	TWR-157	VOL	II	
SEC	200	PAGE	2	



DATE: SUPERSEDES PAGE: DATED: 31 Jul 2000 209-1ff. 30 Jul 1999





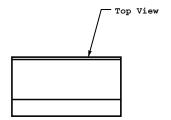
Type I

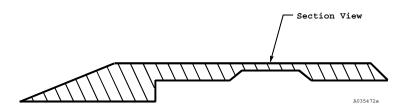
Figure 1. Factory Joint EPDM Weather Seal

A035471a



DATE: 31 Jul 2000 SUPERSEDES PAGE: 209-1ff. DATED: 30 Jul 1999



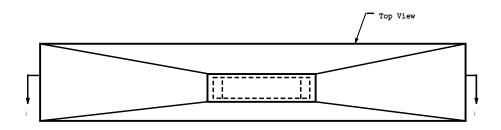


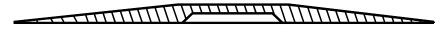
Type II

Figure 2. Systems Tunnel Transition Weather Seal



DATE: SUPERSEDES PAGE: DATED: 31 Jul 2000 209-1ff. 30 Jul 1999





Section A*A Type III

A008846a

Figure 3. Connector Cover EPDM Weather Seal



DATE: SUPERSEDES PAGE: DATED: 31 Jul 2000 209-1ff. 30 Jul 1999

Segment Cross Section

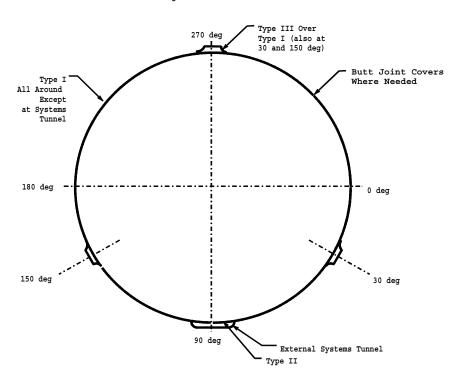
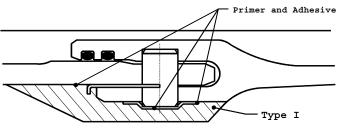


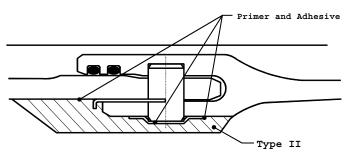
Figure 4. EPDM External Factory Joint Weather Seal Type I, II, III Locations



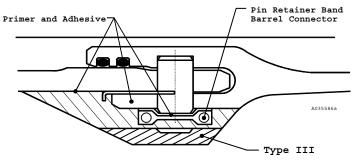
DATE: 31 Jul 2000 SUPERSEDES PAGE: 209-1ff. DATED: 30 Jul 1999







Systems Tunnel Transition Seal



Pin Retainer Band Connector EPDM Weather Seals

Figure 5. External Factory Joint EPDM Weather Seals



DATE: 31 SUPERSEDES PAGE: 200 DATED: 30

31 Jul 2000 209-1ff. 30 Jul 1999

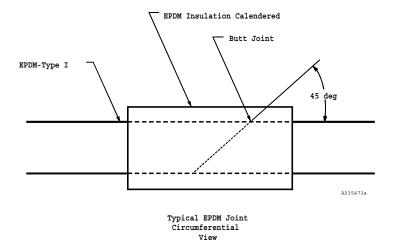


Figure 6. Weather Seal Butt Joint EPDM Cover



DATE: SUPERSEDES PAGE: DATED: 31 Jul 2000

30 Jul 1999

209-1ff.

9.0 RATIONALE FOR RETENTION:

9.1 DESIGN:

DCN FAILURE CAUSES

<u>:N</u>	FAILURE CAUSES		
	A	1.	Formulation from raw materials (base chemicals) of uncured EPDM-neoprene rubber thermal insulation extrusions and calendared stock must meet engineering requirements.
	Α	2.	Raw ingredient properties used in the uncured formula comply with requirements in EPDM, Extrusion, and EPDM Rubber specifications for extruded and calendared stock, respectively, or in EPDM, Silica-Filled, Uncured specifications for both extruded and calendared stock.
	В	3.	EPDM-neoprene rubber extrusions and calendared stock meet the requirements of EPDM, Extrusion or EPDM, Silica-Filled, Uncured specifications regarding configuration and dimensions of extruded stock (Figures 1, 2, 3), and EPDM Rubber or EPDM, Silica-Filled, Uncured specification as to thickness of roll stored sheet and allowable laminae for calendared stock.
	В	4.	Extruded stock and calendared stock must be free from defects that would render them unsuitable for their intended purpose (i.e., voids and inclusions of foreign matter) and be free from contamination per engineering.
	C,D,E,F	5.	Careful engineering review of the manufacturing process is undertaken to avoid contaminating sealant material or surfaces to be sealed and to assure application procedures produce an adequate seal and bond per TWR-16858, TWR-16564, and shop planning.
	C,D,E,F	6.	Engineering specifies contamination control during transportation and handling on raw materials through vendor specification requirements for packaging and handling. Contamination control on vendor-furnished calendared and extruded EPDM stocks is accomplished with appropriate polyethylene film coverings, and on primer and adhesive through best standard commercially available packaging.
	C,D,E,F,G	7.	Contamination control requirements and procedures are described in TWR-16564.
	C,D,E,F	8.	Vacuum bagging is per shop planning.
	C,D,E,F	9.	Allowable vacuum leaks are a controlled process per shop planning.
	G	10.	EPDM optimum cure time and temperature cycles were determined experimentally as documented in TWR-16863.
	A,C,D,E, F,G,H	11.	Witness panels are cured in the autoclave with the insulated segments during the cure cycle. These panels are then tested to assure bondline integrity for primer, adhesive, insulation, liner, and propellant properties were achieved at the end of the cure cycle per engineering, TWR-17123, TWR-64433, and TWR-64923.
	G	12.	Time and temperature cure cycle procedures are per shop planning.
	Н	13.	Physical and mechanical properties of cured EPDM stocks are specified in EPDM,

and calendared stock).

REVISION M

DOC NO.	VOL	II		
SEC	209	PAGE	10	

Extrusion specifications (extruded stock) and EPDM Rubber specifications (calendared stock), or EPDM, Silica-Filled, Uncured specifications (both extruded



		CRITICAL ITEMS LISTS (CILS) No. 10-01-01-10R/01	DATE: SUPERSEDES PAGE: DATED:	31 Jul 2000 209-1ff. 30 Jul 1999
H 1		on adhesive primer and bonding agent lified to engineering requirements.	material properties confo	orm to and
Н 1	15. A struc moisture 17036.	tural safety factor in excess of 2.0 es seal at 33°F in both the horizontal and	exists with the EPDM fa d vertical stacking modes	actory joint per TWR-
H 1	16. Therma 16496.	I analysis of the EPDM factory joint n	noisture seal is covered	per TWR-
l 1	segmen record a	coupling and transportation tests we t per TWR-11712 to verify the adequate actual g-loads during transit. Acceleration were measured and were less than loads.	cy of the tie down provisions of 1.01-g longitudinal	ons and to and 0.86-g
I 1	on the	al tests were accomplished per TWR-12 RSRM forward segment grain. This tion of vibration and shock transportation	testing provided addition	
l 1	transpor Thiokol. handled	ments for handling RSRM componer rtation are similar to those for previo Those requirements dictate that RS by or near a joint to avoid damage. A ety hooks per TWR-13880.	us and other current pr SRM and case segment	rograms at s must be
1 2	weight, RSRM :	cradling or support devices and tie of and contour of components to be tra- segments and other components. Sh are used on trucks and dollies to move	ansported are provided ock mounting and other	to support protective
1 2		re that no damage occurs to flight hard nch site, specially designed 200-ton ra		
1 2	per eng evaluate	transportation shock and vibration lever gineering and loads are derived by ed by Thiokol to verify that shock and ations were not exceeded.	analysis. Monitoring re	ecords are
J,K 2		d rubber has a storage life and temper life may be extended if, after retest, t nents.		
J 2	24. Uncured enginee	d rubber is shipped within the specified ring.	time from date of manu	facture per
J,K 2		d rubber is packaged and shipped in unlight, ultraviolet radiation, ozone, arring.		
J,K 2	a specif	ransportation, the temperature environn ied temperature range. A temperature capable of producing a continuous ring.	recorder is placed with the	ne material



		CRITICAL ITEMS LISTS (CILS) No. 10-01-01-10R/01	DATE: SUPERSEDES PAGE: DATED:	31 Jul 2000 209-1ff. 30 Jul 1999	
J	27.	Selection of material was based upon the rationa rubber on case segments in storage is designe contractual environments, without degradation. 15126, TWR-14669, and TWR-13835.	ed to withstand 5 ye	ars, under	
J	28.	identified. Test conclusions were based on tensil	No significant degradation of EPDM external insulation/case bondline strength was identified. Test conclusions were based on tensile and peel samples taken from aged witness panels as documented in TWR-65281.		
К	29.	Maximum cumulative exposure time out of control engineering.	olled storage is estat	olished per	
К	30.	three types of moisture seals are placed in thin poplaced in containers which are constructed so that	To prevent moisture or fungus degradation during transportation and handling, the three types of moisture seals are placed in thin polyethylene bags. The bags are placed in containers which are constructed so that the extrusions will be adequately protected against moisture or fungus damage per engineering.		
К	31.	EPDM material was shown to be non-nutrient to MIL-STD-810 and TWR-16851.	fungus growth after	testing per	
L	32.	The external factory joint weather seal is compose temperature EPDM-neoprene rubber compound the aerodynamic environment in which it is placed. For and thermal properties are covered in EPDM, ERubber (calendared) or in EPDM, Silica-Filled, Und specifications.	nat was chosen to wit ormulation, physical, n Extrusion (extruded) a	thstand the nechanical, and EPDM	
L	33.	A two-dimensional axisymmetric transient heating program's (ASCHAR) results per TWR-16496 approaching 219°F during ascent aerodynamic h normal seal curing temperature, and pose no se problems that might lead to accelerated ablative we	indicate EPDM ter eating are well below rious chemical bond-	mperatures v even the	
L,M	34.	Wind tunnel tests were performed to verify adequipredicted aeroshear loads per TWR-17243.	uacy of the design to	withstand	
D	35.	A Spray-in-Air cleaning system is used to clean n bonding surface preparation processing sequence.		part of the	
A,B,C,D,E,F G,H,I,L,M	36.	An updated analysis was performed on the fa Performance Enhancement (PE) environments. T PE environment, resulted in stresses essentially idea Aero/Plume Heating Certification. All quoted stressets afety remain unchanged per TWR-66825-3.	his structural analysis entical to those from t	s, using the he Generic	
A,B,C,D,E,F G,H,I,L,M	37.	TWR-61410 was updated to include boundar Performance Enhancement (PE) Program. This conditions created from flight loads. PE temperature temperatures for all locations for the critical time of factory joints and case acreage during flight, tem and maximum case temperatures are lower than a flight load events, PE temperatures are not significant temperatures. There is no impact on pasafety for the case membranes, factory joints, and the same programme to the case membranes.	s report analyzed to ures are equal to curre liftoff. For a few local peratures rise, but of current generic certifical inificantly different from revious analyses or	emperature ent generic tions at the nly slightly, cation. For om current margins of	



CRITICAL ITEMS LISTS (CILS)

No. 10-01-01-10R/01

DATE:

31 Jul 2000

SUPERSEDES PAGE:

209-1ff.

DATED:

30 Jul 1999

9.2 TEST AND INSPECTION:

FAILURE CAUSES and DCN TESTS (T)

CIL CODE

FAA831

 For New EPDM Rubber

A,G	a.	Ingredient properties	ALA026
B,C,D,E,F	b.	Workmanship	ALA039,ALA077
C,D,E,F,K	C.	No shipping or handling damage	ALA044
G	d.	Cured mechanical and physical properties	FAA829
G	e.	Coupon sample is included with shipment for each lot of extrusion	ons ALA007
G,H,L (T)	f.	Ash content	ALA002,ALA003
G,H,L (T)	g.	Hardness	ALA012,ALA013
G,H,L (T)	h.	Initial scorch characteristics	ALA016,ALA017
G,H,L (T)	i.	Mooney viscosity	ALA020,ALA021
G,H,L (T)	j.	Rheology	ALA031,ALA032
G,H,L (T)	k.	Specific gravity	ALA047,ALA048
G,H,L (T)	I.	Tensile strength	ALA060,ALA061
G,H,L (T)	m.	Thermal conductivity	ALA068,ALA069
G,H,L (T)	n.	Ultimate elongation	ALA072,ALA073

2. For Retest EPDM Rubber, verify:

G,H,J a. Chemical and physical properties of EPDM for storage life extension per Table III

3. For New EPDM, Silica-Filled, Uncured, verify:

A,G	a.	Ingredient properties	AMG024R
B,C,D,E,F	b.	Workmanship	AMG038R,AMG074R
C,D,E,F,K	C.	No shipping or handling damage	AMG043R
G,H,L	d.	Vendor records are complete and acceptable	FAA828R
G,H,L (T)	e.	Ash content	AMG002R
G,H,L (T)	f.	Hardness	AMG011R
G,H,L (T)	g.	Initial scorch characteristics	AMG015R
G,H,L (T)	h.	Mooney viscosity	AMG019R
G,H,L (T)	i.	Rheology	AMG030R
G,H,L (T)	j.	Specific gravity	AMG046R
G,H,L (T)	k.	Tensile strength	AMG058R
G,H,L (T)	I.	Ultimate elongation	AMG069R
G,H,L (T)	m.	Tensile adhesion strength	FAA836

4. For Retest EPDM, Silica-Filled, Uncured, verify:

G,H,J	(T)	a.	Tensile strength	FAA832
G,H,J	(T)	b.	Tensile adhesion strength	FAA833
G,H,J	(T)	C.	Ultimate elongation	FAA834

5. For New Adhesive Primer, verify:

Α	(T)	a.	Solids content	AMR059,AMR067
Α	(T)	b.	Density	AMR006,AMR012
Α	(T)	C.	Viscosity	AMR083,AMR092
Α	(T)	d.	Peel adhesion	AMR026,AMR022
Α	. ,	e.	Workmanship	AMR041

REVISION $\underline{\boldsymbol{M}}$

DOC NO.	TWR-157	VOL	II	
SEC	209	PAGE	13	



		CRITICAL ITEMS LISTS (CILS) No. 10-01-01-10R/01	DATE: SUPERSEDES PAGE: DATED:	31 Jul 2000 209-1ff. 30 Jul 1999	
	6.	For New Adhesive, Rubber-to-Metal verify:			
A (T) A (T) A (T) A (T)		 a. Solids content b. Specific gravity c. Viscosity d. Peel strength, rubber-to-steel e. Workmanship For New Insulated Segment Assembly (Forward, Cent 	AND036 AND046 AND014	3,AND026 3,AND036 , AND044 4,AND009 FAA842	
B,G,J,K	••	Storage life and environmental history of EPDM	AFI009A,AMG010R,	ΛΕΙ16 Ω Φ	
			ALA010A,ALA037Q,A		
В		 External insulation defects and repairs are acceptable 	FAA823A,FAA824A	,FAA825A	
C,D,E,F,J C,D,E,F,J		Adhesive primer is used Rubber-to-metal adhesive is used	AMR045,AMR045D,A AKZ024A,FAA838		
C,D,E,F,J C,D,E,F,J C,D,E,F		 e. Storage life is acceptable for adhesive primer f. Storage life of rubber-to-metal adhesive g. Blacklight inspection of joint bonding areas for gr 	AFK185B,AMR048D FAA011,AFE082U	,AMX019	
		oil contamination prior to EPDM insulation lay up	FAA851,FAA85	,	
C,D,E,F C,D,E,F		h. Full coverage of adhesive primeri. Full coverage of adhesive	AFK022,FAA822 AFG024,AFI024E	3,AFK076	
C,D,E,F C,D,E,F		j. Vacuum bags evacuated and checked for leaksk. Case surface meets requirements prior to lay up	AFI160,AFG177 of	7,AFK181	
L		EPDM I. Factory joint weather seal thickness	FAA823,FAA82 AFG128,AFI156		
A,C,D,E F,G,H (T)		m. Results of Chemlok-to-EPDM (weather seal) bon integrity tests with witness panel per engineering	dline		
G		specification n. Air drying of adhesive primer	AOX030,AOX03 AFG057,AFI063		
G G		o. Air drying of rubber-to-metal adhesive p. Insulation cure cycle is complete	AFG058A,AFK239 AFG086,AFI099	,AFI067A	
	8.	For New Case Assembly, Painted Forward Segment, v	verify:		
C,D,E,F,J C,D,E,F,J C,D,E,F G		 a. Storage life is acceptable for adhesive primer b. Adhesive primer is used c. Full coverage of adhesive primer d. Air dry of adhesive primer 		RAA214 RAA218 RAA225 RAA220	
	9. KSC verifies:				
I		 Segments and nozzle components are free of da OMRSD, File V, Vol I, B47SG0.061 	mage per	OMD079	

REVISION M TWR-15712 VOL II

SEC 209 PAGE 14